



Introduction to modelization of thick and heterogeneous plates

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Introduction to modelization of thick and heterogeneous plates

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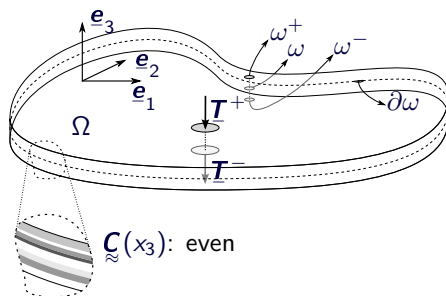
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The 3D Problem

$$\left\{ \begin{array}{l} \sigma_{ij,j}^t = 0 \quad \text{on } \Omega. \\ \sigma_{ij}^t = C_{ijkl}(x_3) \varepsilon_{kl}^t \quad \text{on } \Omega. \\ \sigma_{i3}^t = \pm T_i^\pm \quad \text{on } \omega^\pm. \\ \varepsilon_{ij}^t = u_{(i,j)}^t \quad \text{on } \Omega. \\ u_i^t = 0 \quad \text{on } \partial\omega \times]-t/2, t/2[\end{array} \right.$$

- monoclinic $\underset{\approx}{\mathbf{C}}$:
 $C_{\alpha\beta\gamma 3} = C_{333\alpha} = 0$
- symmetrically laminated plate
- symmetric transverse load
 $\underline{T}^\pm = \frac{p_3}{2} \underline{e}_3$



\Rightarrow pure bending:

- u_3 and $\sigma_{\alpha 3}$ even / x_3
- u_α , $\sigma_{\alpha\beta}$ and σ_{33} odd / x_3

Building a plate model?

For typical width L let $t \rightarrow 0$

- ▶ Solve a 2D problem
- ▶ “fair” 3D displacement localization
- ▶ “fair” 3D stress localization

Some energy principles...

- Statically admissible fields:

$$SA^{3D} = \left\{ \sigma_{ij} \quad / \quad \sigma_{ij,j} = 0 \quad \text{and} \quad \underline{\sigma} \left(\pm \frac{t}{2} \right) \cdot \pm \mathbf{e}_3 = \frac{P_3}{2} \mathbf{e}_3 \right\}$$

$$SA^{3D,0} \Leftrightarrow P_3 = 0$$

- Kinematically admissible fields:

$$KA^{3D} = \left\{ u_i \quad / \quad u_i = 0 \text{ on } \partial\omega \times \left] -\frac{t}{2}, \frac{t}{2} \right[\right\}$$

$$KA^{3D,0} = KA^{3D}$$

- Orthogonality between $KA^{3D,0}$ and $SA^{3D,0}$:

$$\forall \underline{\mathbf{u}} \in KA^{3D,0}, \underline{\sigma} \in SA^{3D,0}, \quad \int_{\Omega} \underline{\sigma} : \underline{\varepsilon}(\underline{\mathbf{u}}) d\Omega = 0$$

Some energy principles...

- Potential energy:

$$\underline{\mathbf{u}}^{3D} = \operatorname{argmin}_{\underline{\mathbf{u}} \in KA^{3D}} \left\{ W^{3D}(\underline{\boldsymbol{\varepsilon}}(\underline{\mathbf{u}})) - \int_{\omega} P_3 \frac{u_3^+ + u_3^-}{2} d\omega \right\}$$

where $W^{3D} = \frac{1}{2} \int_{\Omega} \underline{\boldsymbol{\varepsilon}}(\underline{\mathbf{u}}) : \underline{\mathbf{C}} : \underline{\boldsymbol{\varepsilon}}(\underline{\mathbf{u}}) d\Omega$

- Complementary energy:

$$\underline{\boldsymbol{\sigma}}^{3D} = \operatorname{argmin}_{\underline{\boldsymbol{\sigma}} \in SA^{3D}} \left\{ W^{*3D}(\underline{\boldsymbol{\sigma}}) \right\} \quad \text{where} \quad W^{*3D} = \frac{1}{2} \int_{\Omega} \underline{\boldsymbol{\sigma}} : \underline{\mathbf{S}} : \underline{\boldsymbol{\sigma}} d\Omega$$

$$\underline{\mathbf{S}} = \underline{\mathbf{C}}^{-1}$$

The 2-energy principle

$$\forall \hat{\underline{u}} \in KA^{3D}, \quad \forall \hat{\underline{\sigma}} \in SA^{3D}:$$

$$W^{3D} \left(\underline{\varepsilon}(\hat{\underline{u}}) - \underline{\underline{S}} : \hat{\underline{\sigma}} \right) = W^{*3D} \left(\hat{\underline{\sigma}} - \underline{\underline{C}} : \underline{\varepsilon}(\hat{\underline{u}}) \right)$$

$$= W^{*3D} \left(\hat{\underline{\sigma}} - \underline{\underline{\sigma}}^{3D} \right) + W^{3D} \left(\underline{\varepsilon} \left(\hat{\underline{u}} - \underline{\underline{u}}^{3D} \right) \right)$$

$\Rightarrow W^{*3D} \left(\hat{\underline{\sigma}} - \underline{\underline{C}} : \underline{\varepsilon}(\hat{\underline{u}}) \right)$ provides an error estimate in terms of constitutive equation.

Prager and Synge (1947); Morgenstern (1959); Braess et al. (2010)

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Extension to periodic plates

The case of cellular sandwich panels

Why all periodic plates are not “Reissner” like...

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The case of homogeneous and isotropic plates

The case of laminated plates

Applications

Natural scaling of the stress

$$SA^{3D} \begin{cases} \sigma_{\alpha\beta,\beta} + \sigma_{\alpha 3,3} = 0 \\ \sigma_{\alpha 3,\alpha} + \sigma_{33,3} = 0 \\ \sigma_{33}(\pm t/2) = \pm P_3/2 \\ \sigma_{\alpha 3}(\pm t/2) = 0 \end{cases} \Rightarrow \begin{cases} \sigma_{\alpha 3} = - \int_{-t/2}^{x_3} \sigma_{\alpha\beta,\beta} dz \\ \sigma_{33} = - \int_{t/2}^{x_3} \sigma_{\alpha 3,\alpha} dz - P_3/2 \end{cases}$$

$$\sigma_{\alpha\beta} \sim t^0 \quad \Rightarrow \quad \sigma_{\alpha 3} \sim t^1, \quad \sigma_{33} \sim t^2 \quad \text{and} \quad P_3 \sim t^2$$

From 3D equilibrium to 2D

Plate generalized stresses:

$$\begin{cases} M_{\alpha\beta}(x_1, x_2) = \langle x_3 \sigma_{\alpha\beta} \rangle & \sim t^2 \\ Q_{\alpha}(x_1, x_2) = \langle \sigma_{\alpha 3} \rangle & \sim t^2 \end{cases} \quad \langle \bullet \rangle = \int_{-\frac{t}{2}}^{\frac{t}{2}} \bullet dx_3$$

2D equilibrium equations:

$$\begin{cases} \langle \sigma_{\alpha 3, \alpha} + \sigma_{33, 3} \rangle = 0 \\ \langle x_3 (\sigma_{\alpha\beta, \beta} + \sigma_{\alpha 3, 3}) \rangle = 0 \end{cases} \Rightarrow \begin{cases} Q_{\alpha, \alpha} + P_3 = 0 \\ M_{\alpha\beta, \beta} - Q_{\alpha} = 0 \end{cases}$$

Boussinesq (1871); Mindlin (1951)...

“Kirchhoff’s assumption”

At leading order in t :

$$\begin{cases} \varepsilon_{\alpha\beta} = x_3 K_{\alpha\beta} & \text{where } K_{\alpha\beta} = -U_{3,\alpha\beta} \\ \sigma_{i3} \simeq 0 + O(t^1) & \text{plane stress} \end{cases}$$

$$\varepsilon_{33} \neq 0!!$$

In-plane stress:

$$\sigma_{\alpha\beta} = C_{\alpha\beta\gamma\delta}^{\sigma} \varepsilon_{\delta\gamma} = x_3 C_{\alpha\beta\gamma\delta}^{\sigma} K_{\delta\gamma}$$

where $C_{\alpha\beta\gamma\delta}^{\sigma} = C_{\alpha\beta\gamma\delta} - \frac{C_{\alpha\beta 33} C_{33\gamma\delta}}{C_{3333}}$:

$$\begin{pmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \end{pmatrix} = \frac{E}{1-\nu^2} \begin{pmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{pmatrix} \begin{pmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ 2\varepsilon_{12} \end{pmatrix}$$



The Kirchhoff-Love plate problem

Bending constitutive equation:

$$M_{\alpha\beta} = \langle x_3 \sigma_{\alpha\beta} \rangle = \langle x_3^2 C_{\alpha\beta\gamma\delta}^\sigma \rangle K_{\delta\gamma} = -D_{\alpha\beta\gamma\delta} U_{3,\delta\gamma}$$

$$\underset{\approx}{D} = \frac{t^3}{12} \underset{\approx}{C}^\sigma$$

Statically admissible fields:

$$SA^{KL} : \{ M_{\alpha\beta} / M_{\alpha\beta,\alpha\beta} + P_3 = 0 \}$$

Kinematically compatible fields:

$$KA^{KL} : \{ U_3 / U_3 = 0 \quad \text{and} \quad U_{3,n} = 0 \text{ on } \partial\omega, \underline{n} \text{ outer normal to } \omega \}$$

always "hard support": $U_3 = 0$ on $\partial\omega \Rightarrow U_{3,t} = 0$ on $\partial\omega$

Building SA^{3D} fields

$$\hat{\sigma}_{\alpha\beta} = x_3 C_{\alpha\beta\gamma\delta}^{\sigma} d_{\delta\gamma\epsilon\zeta} M_{\zeta\epsilon} = \frac{12x_3}{t^3} M_{\alpha\beta} \sim t^0$$

$$\hat{\sigma}_{\alpha 3} = - \int_{-t/2}^{x_3} \sigma_{\alpha\beta,\beta} dz = \frac{3}{2t} \left(1 - \frac{4x_3^2}{t^2} \right) M_{\alpha\beta,\beta} \sim t^1$$

$$\hat{\sigma}_{33} = - \int_{-t/2}^{x_3} \sigma_{\alpha 3,\alpha} dz - P_3/2 = \frac{3x_3}{2t} \left(1 - \frac{4x_3^2}{3t^2} \right) P_3 \sim t^2$$

$$\Leftrightarrow \hat{\underline{\underline{\sigma}}} = \underline{\underline{\mathbb{S}}}^M(x_3) : \underline{\underline{M}} + \underline{\underline{\mathbb{S}}}^Q(x_3) \cdot (\underline{\underline{M}} \cdot \underline{\underline{\nabla}}) + \underline{\underline{\mathbb{S}}}^{P_3}(x_3) P_3$$

$$\hat{\underline{\underline{\sigma}}} \in SA^{3D}$$

Building KA^{3D} fields

Strains

$$\hat{\epsilon} = \begin{cases} \hat{\epsilon}_{\alpha\beta} &= x_3 d_{\alpha\beta\gamma\delta} M_{\gamma\delta} &= S_{\alpha\beta\gamma\delta} \hat{\sigma}_{\delta\gamma} + \cancel{S_{\alpha\beta 33} \hat{\sigma}_{33}} \\ \hat{\epsilon}_{\alpha 3} &= \frac{3}{4Gt} \left(1 - \frac{4x_3^2}{t^2}\right) M_{\alpha\beta,\beta} &= 2S_{\alpha 3\beta 3} \hat{\sigma}_{\beta 3} \\ \hat{\epsilon}_{33} &= -\frac{12\nu x_3}{Et^3} M_{\alpha\alpha} &= S_{33\alpha\beta} \hat{\sigma}_{\alpha\beta} + \cancel{S_{33 33} \hat{\sigma}_{33}} \end{cases}$$

Building KA^{3D} fields

Integration

$$\hat{u}_3 = \int^{x_3} \hat{\varepsilon}_{33}(z) dz + U_3 = \underbrace{\mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha}}_{\sim t^1} + \underbrace{U_3}_{\sim t^{-1}}$$

$$\text{where: } \mathbb{U}_{3\alpha\beta}^M(x_3) = \frac{\nu}{2Et} \left(\frac{12x_3^2}{t^2} - 1 \right) \delta_{\alpha\beta} \quad \text{and} \quad \langle \mathbb{U}_{3\alpha\beta}^M(x_3) \rangle = 0$$

Building KA^{3D} fields

Integration

$$\hat{u}_3 = \int^{x_3} \hat{\varepsilon}_{33}(z) dz + U_3 = \underbrace{\mathfrak{u}_{3\alpha\beta}^M(x_3) M_{\beta\alpha}}_{\sim t^1} + \underbrace{U_3}_{\sim t^{-1}}$$

$$\hat{u}_\alpha = \int^{x_3} 2\hat{\varepsilon}_{\alpha 3}(z) - \hat{u}_{3,\alpha} dz = \underbrace{\mathfrak{u}_{\alpha\beta\gamma\delta}^{M\otimes\nabla}(x_3) M_{\delta\gamma,\beta}}_{\sim t^2} - \underbrace{x_3 U_{3,\alpha}}_{\sim t^0}$$

where:

$$\mathfrak{u}_{\alpha\beta\gamma\delta}^{M\otimes\nabla}(x_3) M_{\delta\gamma\beta} = \frac{x_3}{2Et} \left(6(1+\nu) \left(1 - \frac{4x_3^2}{3t^2} \right) M_{\alpha\beta,\beta} + \nu \left(1 - \frac{4x_3^2}{t^2} \right) M_{\beta\beta,\alpha} \right)$$

$$\text{and } \left\langle \mathfrak{u}_{\alpha\beta\gamma\delta}^{M\otimes\nabla}(x_3) \right\rangle = 0$$

Building KA^{3D} fields

Integration

$$\hat{u}_3 = \int^{x_3} \hat{\varepsilon}_{33}(z) dz + U_3 = \underbrace{\mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha}}_{\sim t^1} + \underbrace{U_3}_{\sim t^{-1}}$$

$$\hat{u}_\alpha = \int^{x_3} 2\hat{\varepsilon}_{\alpha 3}(z) - \hat{u}_{3,\alpha} dz = \underbrace{\mathbb{U}_{\alpha\beta\gamma\delta}^{M\otimes\nabla}(x_3) M_{\delta\gamma,\beta}}_{\sim t^2} - \underbrace{x_3 U_{3,\alpha}}_{\sim t^0}$$

$$\Rightarrow \quad \hat{\underline{u}} = U_3 \underline{e}_3 - x_3 U_{3,\alpha} \underline{e}_\alpha + \mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha} \underline{e}_3 + \mathbb{U}_{\alpha\beta\gamma\delta}^{M\otimes\nabla}(x_3) M_{\delta\gamma,\beta} \underline{e}_\alpha$$

with:

$$\hat{\underline{\varepsilon}} = \underline{\varepsilon}(\hat{\underline{u}})$$

Application of the Two-Energy principle

Consider U_3^{KL} and $\tilde{\mathbf{M}}^{KL}$ the solution of the Kirchhoff-Love plate problem and define:

$$\underline{\sigma}^{KL} = \underline{\mathbb{S}}^M(x_3) : \tilde{\mathbf{M}}^{KL} + \underline{\mathbb{S}}^Q(x_3) \cdot \left(\tilde{\mathbf{M}}^{KL} \cdot \underline{\nabla} \right) + \underline{\mathbb{S}}^{P_3}(x_3) P_3$$

$$\underline{\mathbf{u}}^{KL} = U_3^{KL} \mathbf{e}_3 - x_3 U_{3,\alpha}^{KL} \mathbf{e}_\alpha + \mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha}^{KL} \mathbf{e}_3 + \mathbb{U}_{\alpha\beta\gamma\delta}^{M \otimes \nabla}(x_3) M_{\delta\gamma,\beta}^{KL} \mathbf{e}_\alpha$$

We have:

$$\underline{\varepsilon}(\underline{\mathbf{u}}^{KL}) - \underline{\mathbb{S}} : \underline{\sigma}^{KL} = \underline{\mathbb{S}}_{33}^{P_3} P_3 \begin{pmatrix} S_{\alpha\beta 33} & 0 \\ 0 & 0 & S_{3333} \end{pmatrix} \sim t^2$$

Application of the Two-Energy principle

$$\underline{\underline{\varepsilon}}(\underline{\underline{u}}^{KL}) - \underline{\underline{\mathbf{S}}} : \underline{\underline{\sigma}}^{KL} \sim t^2 \quad \Rightarrow \quad W^{3D} \left(\underline{\underline{\varepsilon}}(\underline{\underline{u}}^{KL}) - \underline{\underline{\mathbf{S}}} : \underline{\underline{\sigma}}^{KL} \right) \sim t^5$$

Would lead to a relative error in t^2 ...

$$\underline{\underline{\sigma}}^{KL} \in SA^{3D} \quad \text{but} \quad \underline{\underline{u}}^{KL} \notin KA^{3D}$$

At best: relative error in $t^{1/2}$...

Braess et al. (2010) among many others!

Reissner's original plate model (1945)

min of W^{*3D}

Let us consider:

$$\hat{\underline{\sigma}} = \underline{\mathbb{s}}^M(x_3) : \underline{\underline{M}} + \underline{\mathbb{s}}^Q(x_3) \cdot \underline{\underline{Q}} + \underline{\mathbb{s}}^{P_3}(x_3) P_3$$

with

$$SA^{RM} = \left\{ (\underline{\underline{M}}, \underline{\underline{Q}}) / Q_{\alpha,\alpha} + P_3 = 0 \quad \text{and} \quad M_{\alpha\beta,\beta} - Q_\alpha = 0 \quad \text{on} \quad \omega \right\}$$

$$W^{*3D}(\underline{\underline{\sigma}}^{3D}) \leq \min_{(\underline{\underline{M}}, \underline{\underline{Q}}) \in SA^{RM}} W^{*3D}(\hat{\underline{\underline{\sigma}}}) \leq W^{*3D}(\underline{\underline{\sigma}}^{KL})$$

is a better approximation of $W^{*3D}(\underline{\underline{\sigma}}^{3D})$

Reissner's original plate model (1945)

Dualization

$$\left\{ \begin{array}{ll} Q_{\alpha,\alpha} + P_3 = 0 & \times U_3 \\ M_{\alpha\beta,\beta} - Q_\alpha = 0 & \times \varphi_\alpha \end{array} \right. \Rightarrow \left\{ \begin{array}{lll} Q_\alpha & \leftrightarrow & \gamma_\alpha = \varphi_\alpha + U_{3,\alpha} \\ M_{\alpha\beta} & \leftrightarrow & \chi_{\alpha\beta} = \varphi_{(\alpha,\beta)} \end{array} \right.$$

$$KA^{RM} = \{(U_3, \varphi_\alpha) / U_3 = 0 \text{ and } \varphi_\alpha = 0 \text{ on } \partial\omega\}$$

Reissner's original plate model (1945)

Constitutive equation

$$\begin{aligned}
 W^{*RM}(\underline{\tilde{M}}, \underline{\tilde{Q}}) &= W^{*3D}(\hat{\sigma}) \\
 &= \frac{1}{2} \int_{\omega} \begin{pmatrix} \underline{\tilde{M}} \\ \underline{\tilde{Q}} \\ P_3 \end{pmatrix}^T \begin{pmatrix} \underline{\tilde{d}} & 0 & \frac{6\nu}{5Et} \underline{\tilde{\delta}} \\ 0 & \frac{6}{5Gt} \underline{\tilde{\delta}} & 0 \\ \frac{6\nu}{5Et} \underline{\tilde{\delta}} & 0 & \frac{17t}{140E} \end{pmatrix} \begin{pmatrix} \underline{\tilde{M}} \\ \underline{\tilde{Q}} \\ P_3 \end{pmatrix} d\omega \\
 &\Rightarrow \begin{cases} \underbrace{\chi_{\alpha\beta}}_{\sim t^{-1}} = \underbrace{d_{\alpha\beta\gamma\delta} M_{\delta\gamma}}_{\sim t^{-1}} + \underbrace{\frac{6\nu}{5Et} \delta_{\alpha\beta} P_3}_{\sim t^1} \\ \gamma_{\alpha} = \frac{6}{5Gt} Q_{\alpha} \sim t^1 \end{cases}
 \end{aligned}$$

The contribution of P_3 is almost always dropped in the literature



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Field Localization

Following the same procedure leads to:

$$\hat{\underline{\sigma}} = \underline{\mathbb{S}}^M(x_3) : \underline{\mathbb{M}} + \underbrace{\underline{\mathbb{S}}^R(x_3) : \underline{\mathbb{M}} \otimes \underline{\nabla}}_{\sim t^1} + \underbrace{\underline{\mathbb{S}}^T(x_3) : \underline{\mathbb{M}} \otimes \underline{\nabla}^2 + \underline{\mathbb{S}}^{P_3}(x_3) P_3}_{\sim t^2}$$

$$\hat{\underline{u}} = U_3 \underline{e}_3 - x_3 U_{3,\alpha} \underline{e}_\alpha + \mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha} \underline{e}_3 + \mathbb{U}_{\alpha\beta\gamma\delta}^R(x_3) M_{\delta\gamma,\beta} \underline{e}_\alpha$$

where $\underline{\mathbb{R}} = \underline{\mathbb{M}} \otimes \underline{\nabla}$ and $\underline{\mathbb{T}} = \underline{\mathbb{R}} \otimes \underline{\nabla}$

... Kirchhoff-Love error estimates still hold.

Building SA^{3D} fields

$$\hat{\sigma}_{\alpha\beta} = x_3 C_{\alpha\beta\gamma\delta}^{\sigma} d_{\delta\gamma\epsilon\zeta} M_{\zeta\epsilon} = \mathbb{S}_{\alpha\beta\gamma\delta}^M(x_3) M_{\delta\gamma}$$

$$\hat{\sigma}_{\alpha 3} = - \int_{-t/2}^{x_3} \sigma_{\alpha\beta,\beta} dz = \mathbb{S}_{\alpha 3\beta\gamma\delta}^R(x_3) M_{\delta\gamma,\beta}$$

$$\hat{\sigma}_{33} = - \int_{-t/2}^{x_3} \sigma_{\alpha 3,\alpha} dz - P_3/2 = \mathbb{S}_{33\alpha\beta\gamma\delta}^T(x_3) M_{\delta\gamma,\beta\alpha} + \mathbb{S}_{33}^{P_3}(x_3) P_3$$

The Bending-Gradient constitutive equation

Extending Reissner's approach?:

$$\hat{\underline{\underline{\sigma}}} = \underline{\underline{\mathbf{s}}}^M(x_3) : \underline{\underline{\mathbf{M}}} + \underline{\underline{\mathbf{s}}}^R(x_3) : \underline{\underline{\mathbf{R}}} + \underline{\underline{\mathbf{s}}}^T(x_3) : \underline{\underline{\mathbf{T}}} + \underline{\underline{\mathbf{s}}}^{P_3}(x_3) P_3$$

Let us define:

$$\hat{\underline{\underline{\sigma}}}^* = \underline{\underline{\mathbf{s}}}^M(x_3) : \underline{\underline{\mathbf{M}}} + \underline{\underline{\mathbf{s}}}^R(x_3) : \underline{\underline{\mathbf{R}}}$$

with: $\hat{\underline{\underline{\sigma}}}^* \cdot \underline{\underline{\nabla}} = 0 + O(t^1)$ only.

$$W^{*BG}(\underline{\underline{\mathbf{M}}}, \underline{\underline{\mathbf{R}}}) = W^{*3D}(\hat{\underline{\underline{\sigma}}}^*) = \frac{1}{2} \int_{\omega} \left(\underline{\underline{\mathbf{M}}} : \underline{\underline{\mathbf{d}}} : \underline{\underline{\mathbf{M}}} + \underline{\underline{\mathbf{R}}} : \underline{\underline{\mathbf{f}}} : \underline{\underline{\mathbf{R}}} \right) d\omega$$

The Bending-Gradient theory for thick plates

- Equilibrium equations:

$$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 \end{cases}$$

$$\text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} \\ Q_{\alpha,\alpha} + P_3 = 0 \end{cases}$$

- Mechanical meaning of \underline{R}

$$Q_\alpha = R_{\alpha\beta\beta} \Leftrightarrow \begin{cases} Q_1 = R_{111} + R_{122} = M_{11,1} + M_{12,2} \\ Q_2 = R_{121} + R_{222} = M_{21,1} + M_{22,2} \end{cases}$$

The Bending-Gradient theory for thick plates

► Equilibrium equations:

$$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} & \times \Phi_{\alpha\beta\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 & \times U_3 \end{cases}$$

$$\text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} & \times \varphi_\alpha \\ Q_{\alpha,\alpha} + P_3 = 0 & \times U_3 \end{cases}$$

The Bending-Gradient theory for thick plates

► Equilibrium equations:

$$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} & \times \Phi_{\alpha\beta\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 & \times U_3 \end{cases} \quad \text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} & \times \varphi_\alpha \\ Q_{\alpha,\alpha} + P_3 = 0 & \times U_3 \end{cases}$$

	Equilibrium	Work of internal forces	Work on Boundary
$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \Phi_{\alpha\beta\epsilon,\epsilon} +$ $R_{\alpha\beta\gamma} (\Phi_{\alpha\beta\gamma} + I_{\alpha\beta\gamma\epsilon} U_{3,\epsilon})$	$M_{\alpha\beta} \Phi_{\alpha\beta\gamma} n_\gamma +$ $R_{\alpha\beta\beta} n_\alpha U_3$	
$\text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} \\ Q_{\alpha,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \varphi_{(\alpha,\beta)} +$ $Q_\alpha (\varphi_\alpha + U_{3,\alpha})$	$M_{\alpha\beta} n_\beta \varphi_\alpha +$ $Q_\alpha n_\alpha U_3$	

$$I_{\alpha\beta\gamma\delta} = \frac{1}{2} (\delta_{\alpha\gamma} \delta_{\beta\delta} + \delta_{\alpha\delta} \delta_{\beta\gamma}) \text{ identity for in-plane elasticity}$$

The Bending-Gradient theory for thick plates

► Equilibrium equations:

$$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} & \times \Phi_{\alpha\beta\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 & \times U_3 \end{cases} \quad \text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} & \times \varphi_\alpha \\ Q_{\alpha,\alpha} + P_3 = 0 & \times U_3 \end{cases}$$

	Equilibrium	Work of internal forces	Work on Boundary
BG:	$\begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \Phi_{\alpha\beta\epsilon,\epsilon} + R_{\alpha\beta\gamma} (\Phi_{\alpha\beta\gamma} + I_{\alpha\beta\gamma\epsilon} U_{3,\epsilon})$	$M_{\alpha\beta} \Phi_{\alpha\beta\gamma} n_\gamma + R_{\alpha\beta\beta} n_\alpha U_3$
RM:	$\begin{cases} Q_\alpha = M_{\alpha\beta,\beta} \\ Q_{\alpha,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \varphi_{(\alpha,\beta)} + Q_\alpha (\varphi_\alpha + U_{3,\alpha})$	$M_{\alpha\beta} n_\beta \varphi_\alpha + Q_\alpha n_\alpha U_3$

When the plate is homogeneous, BG is exactly turned into RM
(loss of invertibility of $\underline{\underline{f}}$)

The Bending-Gradient theory for thick plates

► Equilibrium equations:

$$\text{BG: } \begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} & \times \Phi_{\alpha\beta\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 & \times U_3 \end{cases} \quad \text{RM: } \begin{cases} Q_\alpha = M_{\alpha\beta,\beta} & \times \varphi_\alpha \\ Q_{\alpha,\alpha} + P_3 = 0 & \times U_3 \end{cases}$$

	Equilibrium	Work of internal forces	Work on Boundary
BG:	$\begin{cases} R_{\alpha\beta\gamma} = M_{\alpha\beta,\gamma} \\ R_{\alpha\beta\beta,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \Phi_{\alpha\beta\epsilon,\epsilon} + R_{\alpha\beta\gamma} (\Phi_{\alpha\beta\gamma} + I_{\alpha\beta\gamma\epsilon} U_{3,\epsilon})$	$M_{\alpha\beta} \Phi_{\alpha\beta\gamma} n_\gamma + R_{\alpha\beta\beta} n_\alpha U_3$
RM:	$\begin{cases} Q_\alpha = M_{\alpha\beta,\beta} \\ Q_{\alpha,\alpha} + P_3 = 0 \end{cases}$	$M_{\alpha\beta} \varphi_{(\alpha,\beta)} + Q_\alpha (\varphi_\alpha + U_{3,\alpha})$	$M_{\alpha\beta} n_\beta \varphi_\alpha + Q_\alpha n_\alpha U_3$

Low order interpolation for U_3 and Φ !

Local Fields reconstruction

Once the plate problem is solved (U_3^{BG} , Φ^{BG} , \tilde{M}^{BG} , \tilde{R}^{BG} known), we suggest the following field reconstruction:

$$\blacktriangleright \underline{\sigma}^{BG} = \underline{\mathbb{S}}^M(x_3) : \underline{M} + \underline{\mathbb{S}}^R(x_3) : \underline{R} + \underline{\mathbb{S}}^T(x_3) : \underline{R} \otimes \underline{\nabla} + \underline{\mathbb{S}}^{P_3}(x_3) P_3$$

$$\blacktriangleright \underline{u}^{BG} = U_3 \underline{e}_3 - x_3 U_{3,\alpha} \underline{e}_\alpha + \mathbb{U}_{3\alpha\beta}^M(x_3) M_{\beta\alpha} \underline{e}_3 + \mathbb{U}_{\alpha\beta\gamma\delta}^R(x_3) R_{\delta\gamma\beta} \underline{e}_\alpha$$

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Cylindrical bending of laminates

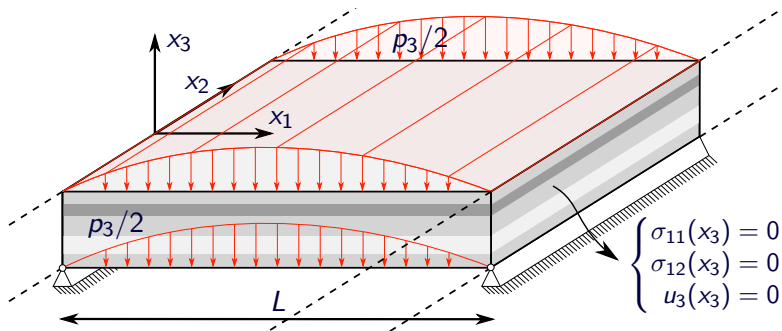
Extension to periodic plates

The case of cellular sandwich panels

Why all periodic plates are not “Reissner” like...

Pagano's boundary value problem (Pagano, 1969)

CFRP layers with different orientations:



Invariant in x_2 -Direction, "periodic" in x_1 -Direction
 \Rightarrow No boundary layer!

Practical Localization...

Kirchhoff-Love

$$\blacktriangleright \underline{\sigma}^{KL} = \underline{\mathbb{s}}^M(x_3) : \underline{M}^{KL} + \underline{\mathbb{s}}^Q(x_3) \cdot (\underline{M}^{KL} \cdot \underline{\nabla}) + \underline{\mathbb{s}}^{P_3}(x_3) P_3$$

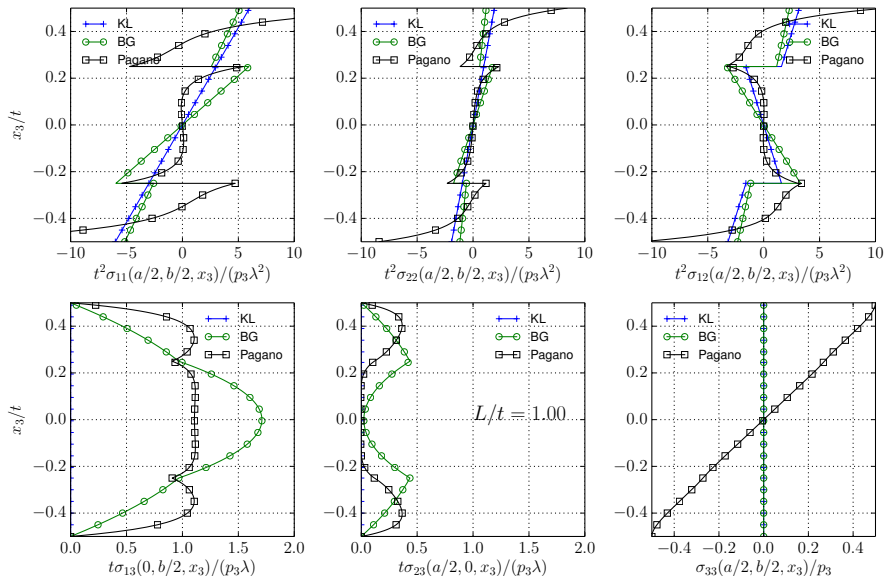
$$\blacktriangleright \underline{u}^{KL} = U_3^{KL} \mathbf{e}_3 - x_3 U_{3,\alpha}^{KL} \mathbf{e}_\alpha + \mathbb{u}_{3\alpha\beta}^M(x_3) M_{\beta\alpha}^{KL} \mathbf{e}_3 + \mathbb{u}_{\alpha\beta\gamma\delta}^{M \otimes \nabla}(x_3) M_{\delta\gamma,\beta}^{KL} \mathbf{e}_\alpha$$

Bending-Gradient

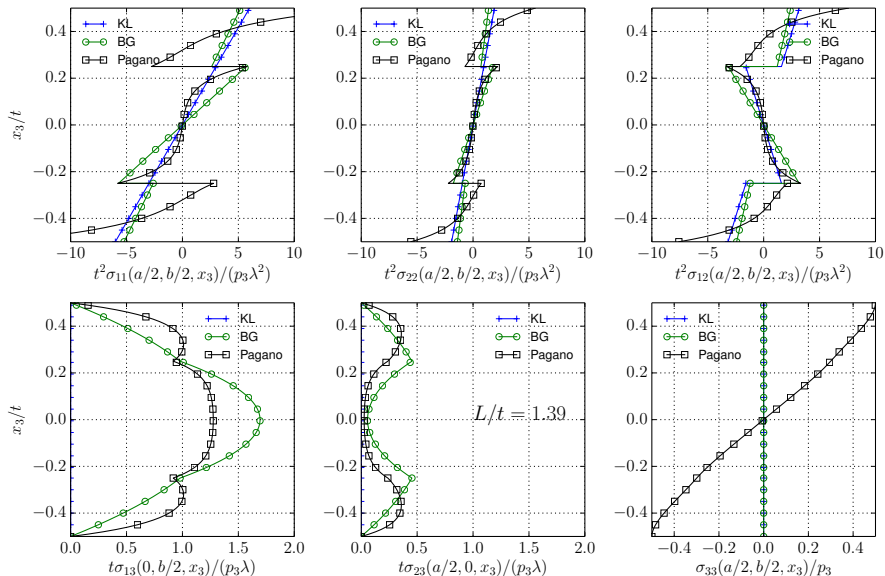
$$\blacktriangleright \underline{\sigma}^{BG} = \underline{\mathbb{s}}^M(x_3) : \underline{M} + \underline{\mathbb{s}}^R(x_3) : \underline{R} + \underline{\mathbb{s}}^T(x_3) : \underline{R} \otimes \underline{\nabla} + \underline{\mathbb{s}}^{P_3}(x_3) P_3$$

$$\blacktriangleright \underline{u}^{BG} = U_3 \mathbf{e}_3 - x_3 U_{3,\alpha} \mathbf{e}_\alpha + \mathbb{u}_{3\alpha\beta}^M(x_3) M_{\beta\alpha} \mathbf{e}_3 + \mathbb{u}_{\alpha\beta\gamma\delta}^R(x_3) R_{\delta\gamma\beta} \mathbf{e}_\alpha$$

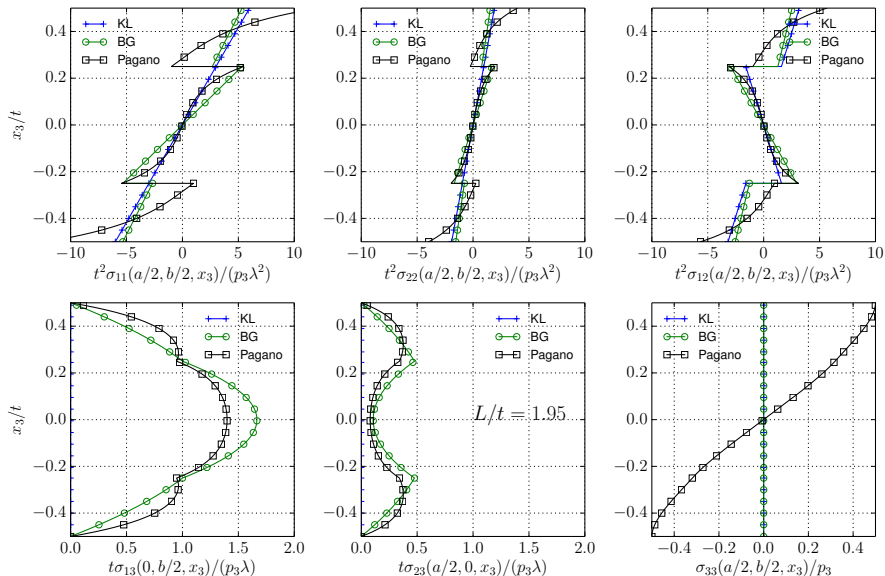
Stress distributions for a $[30^\circ, -30^\circ, 30^\circ]$ stack



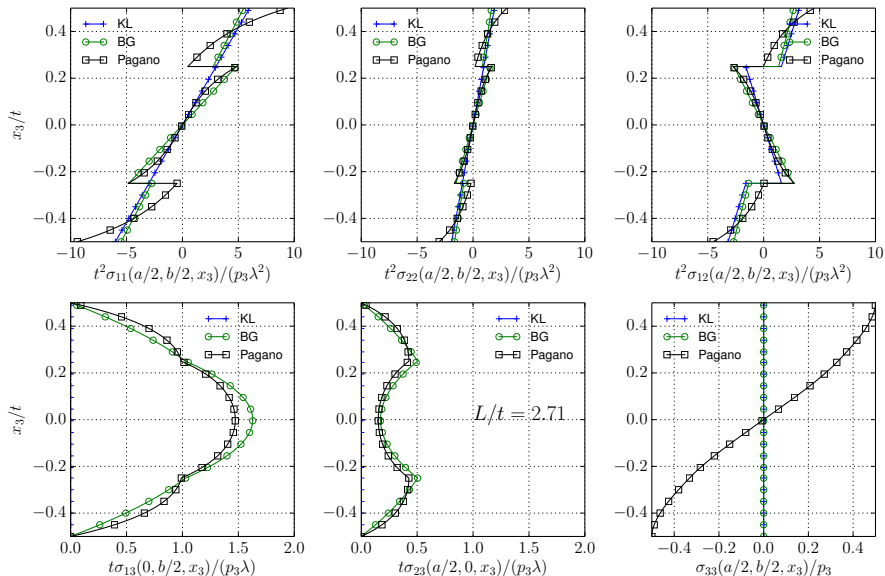
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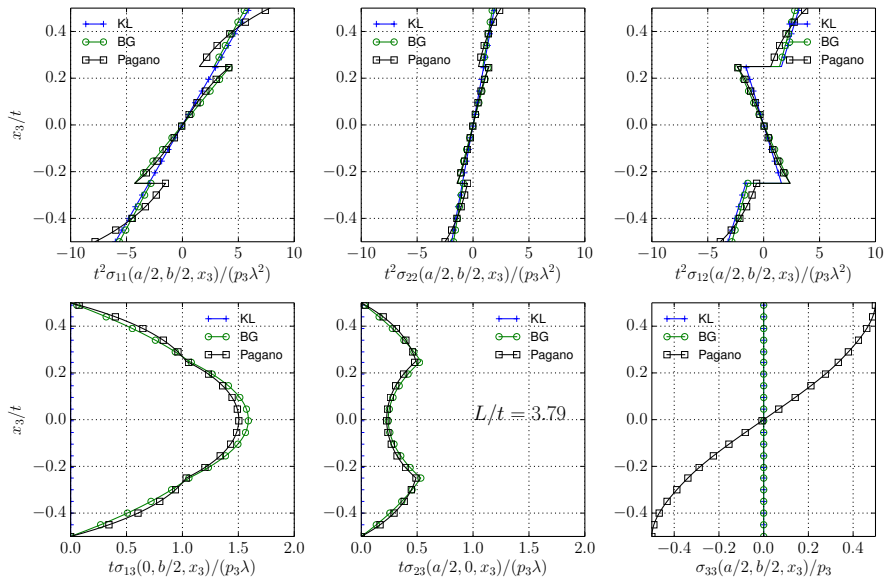
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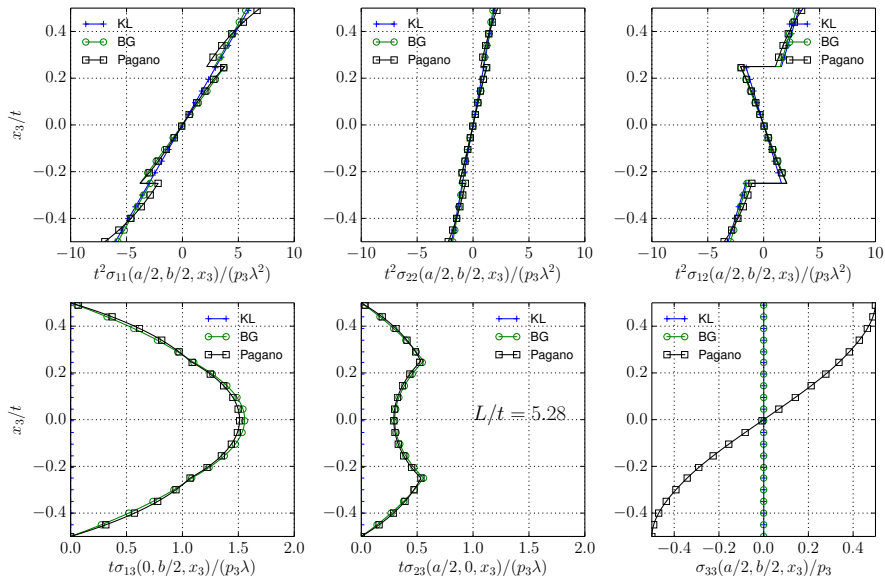
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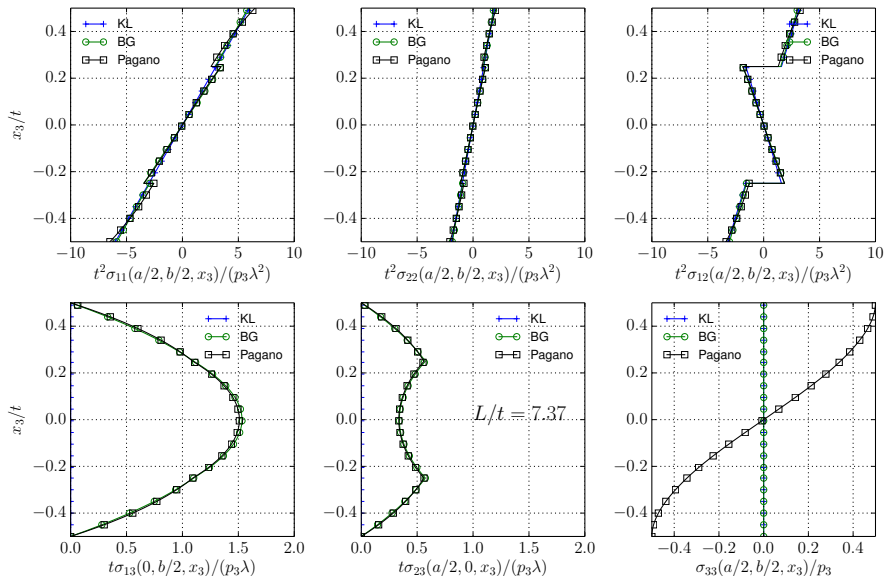
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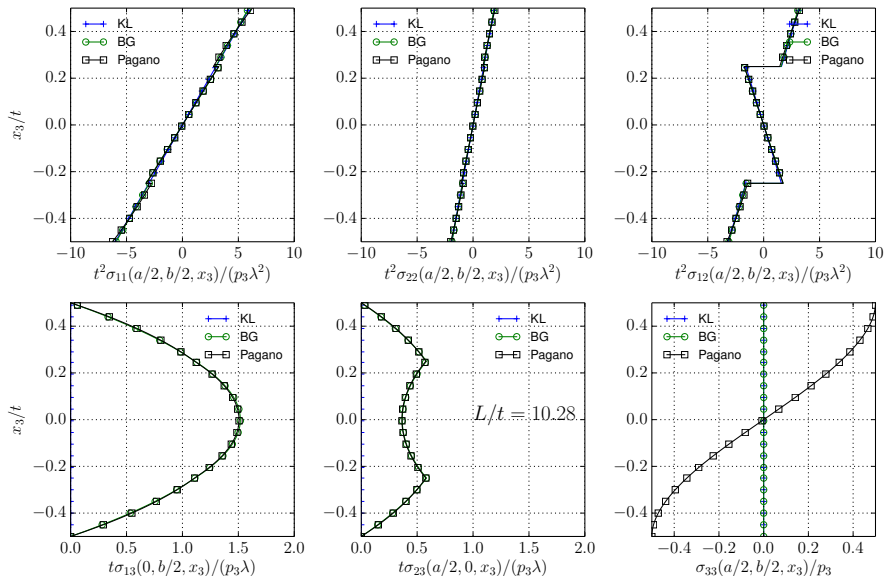
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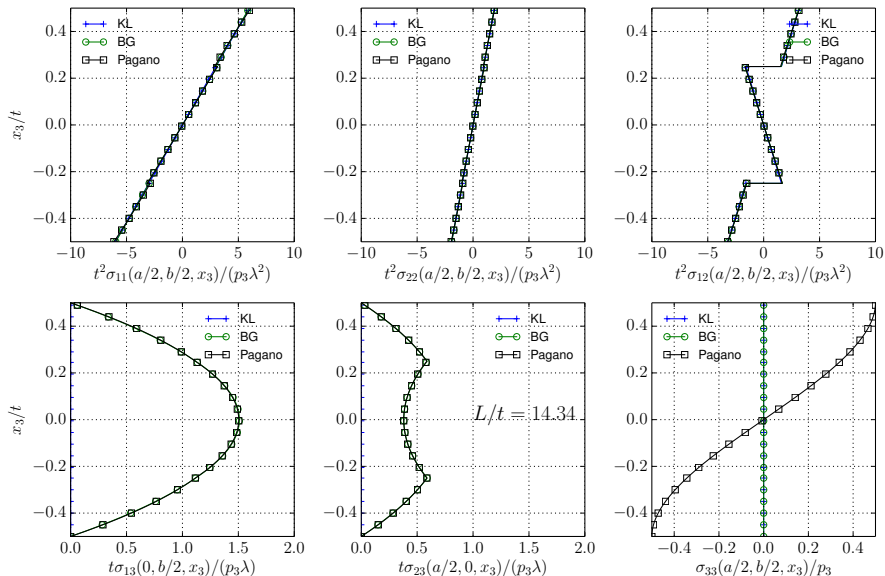
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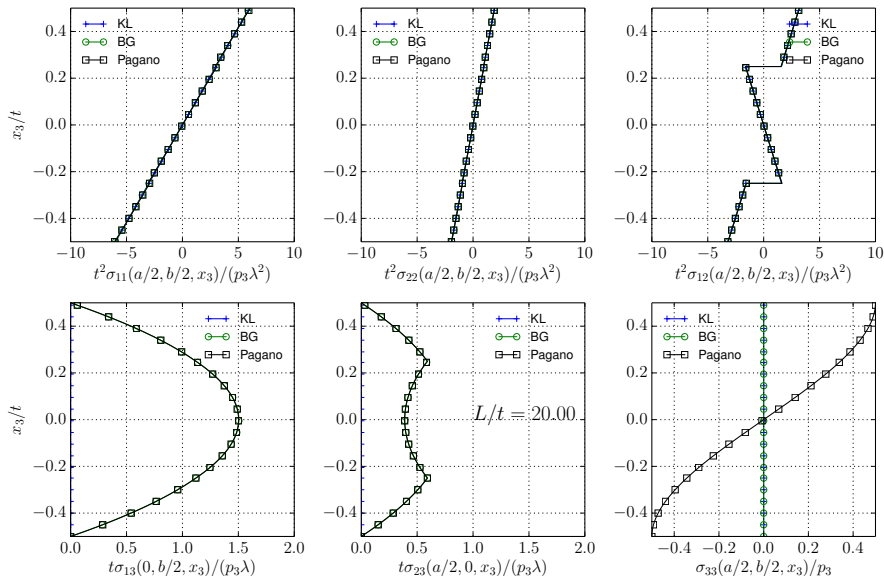
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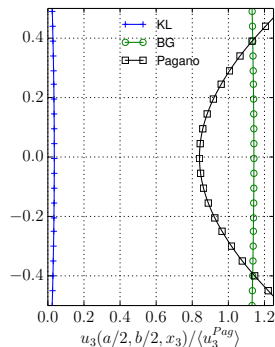
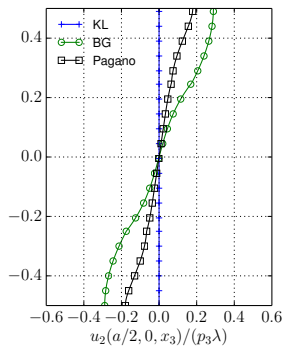
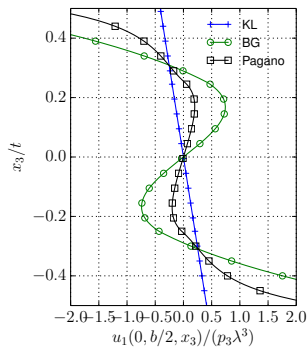
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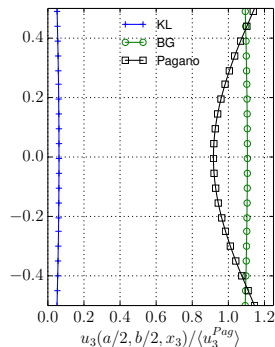
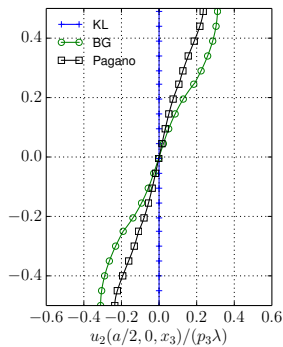
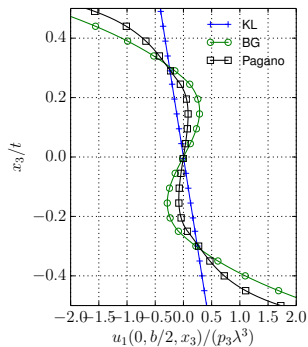
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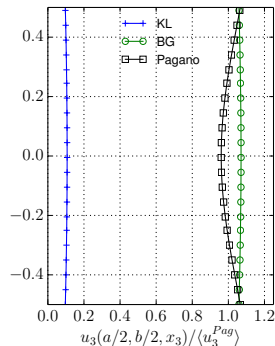
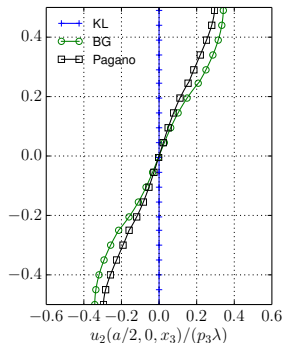
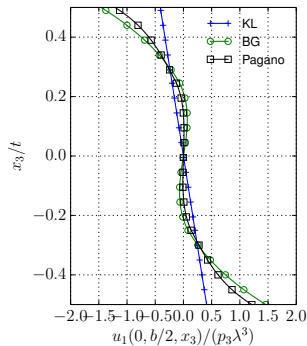
Displacement distributions for a $[30^\circ, -30^\circ, 30^\circ]$ stack



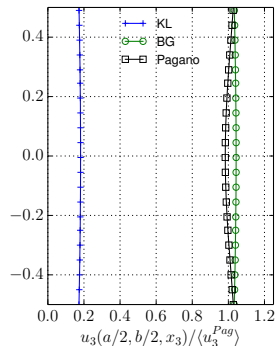
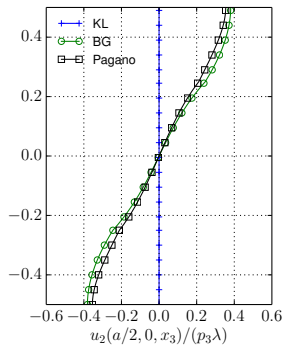
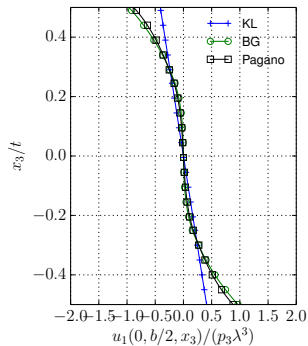
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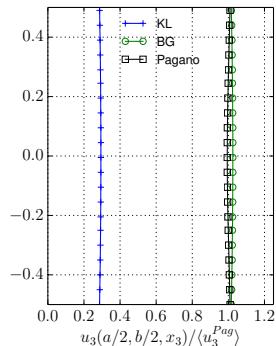
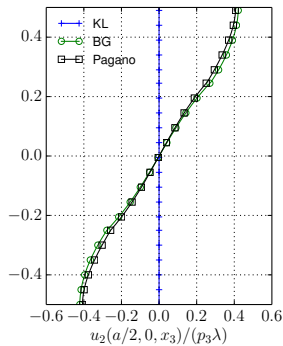
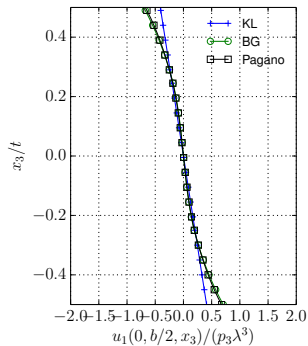
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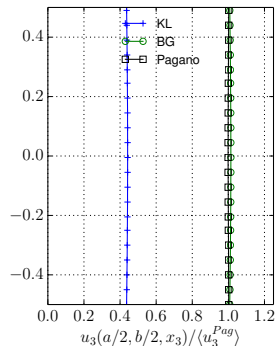
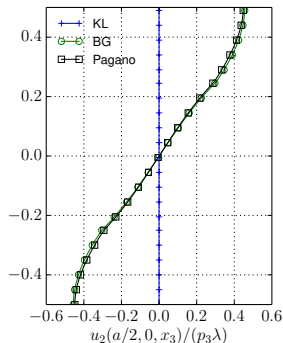
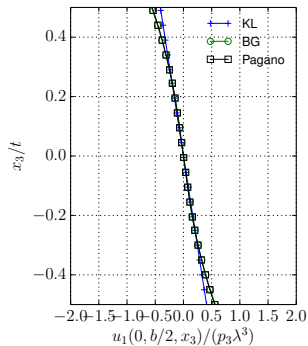
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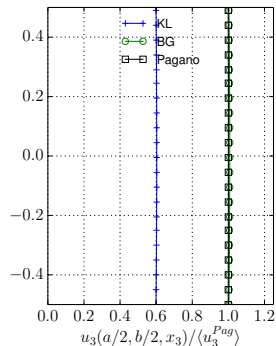
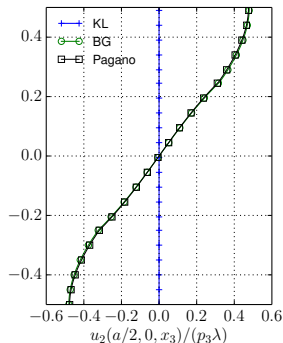
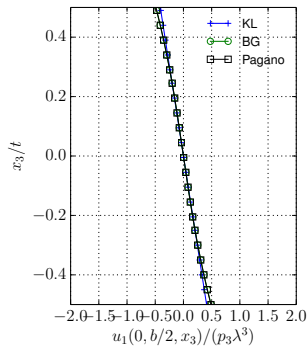
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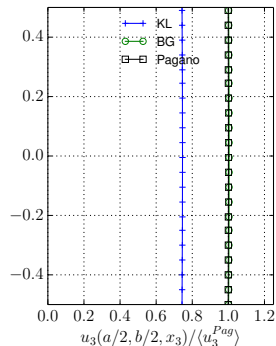
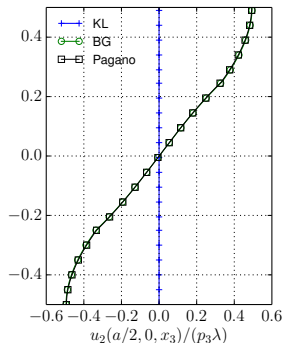
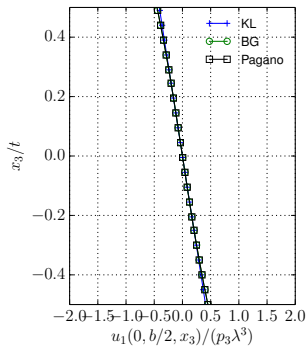
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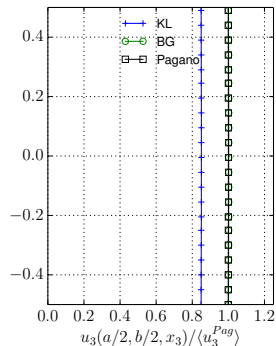
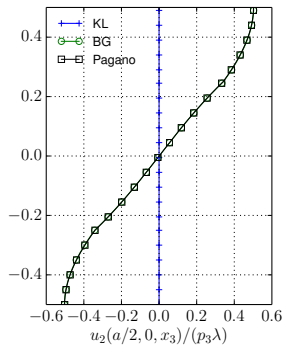
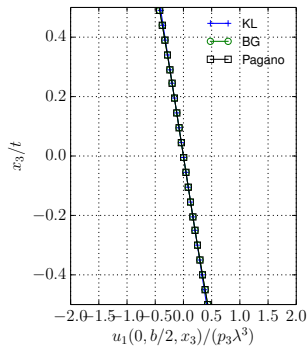
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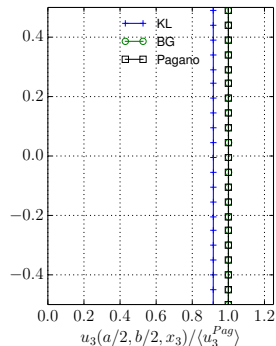
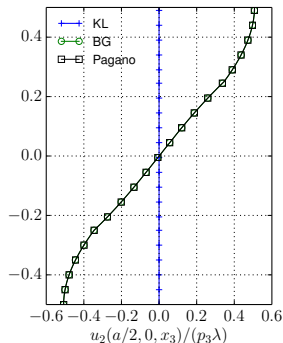
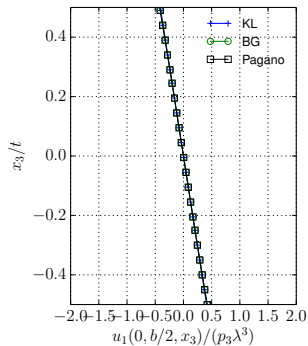
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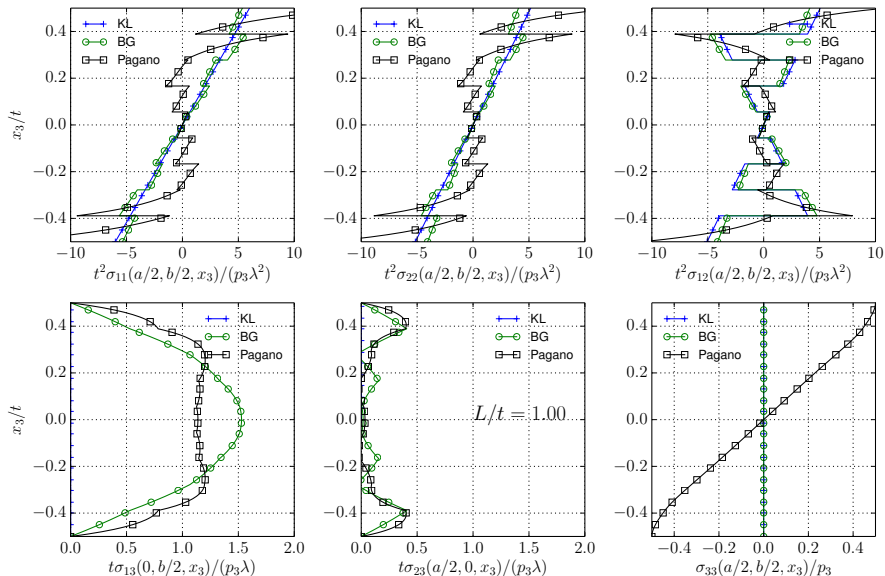
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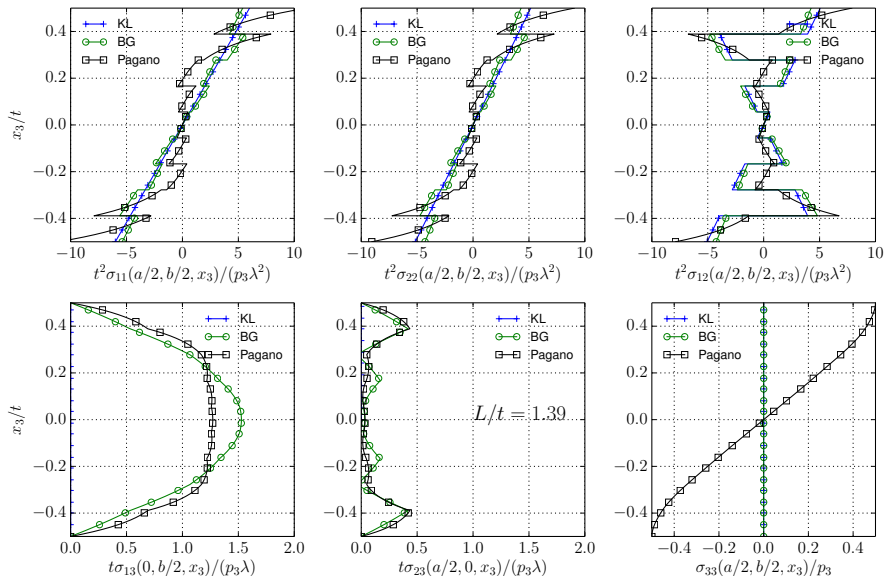
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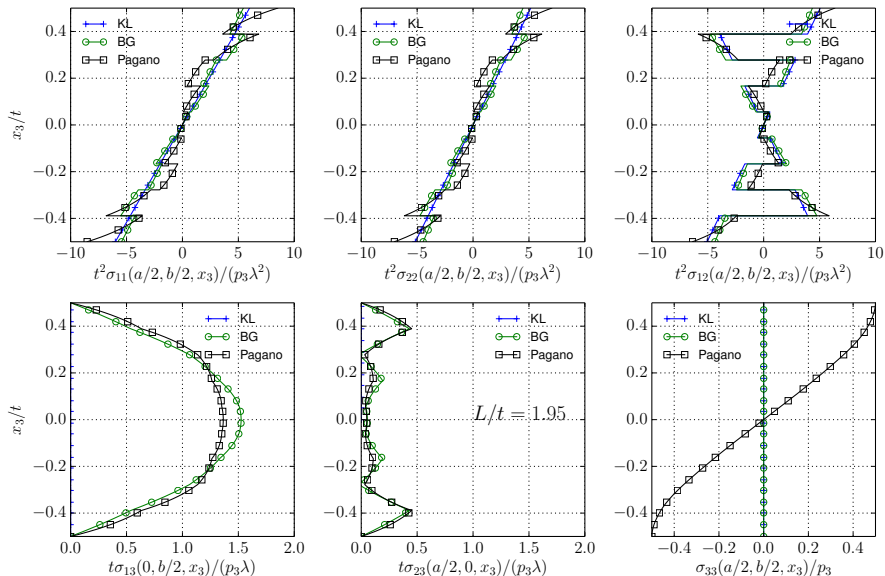
Stress distributions for a $[45^\circ, -45^\circ]^4, 45^\circ$ stack



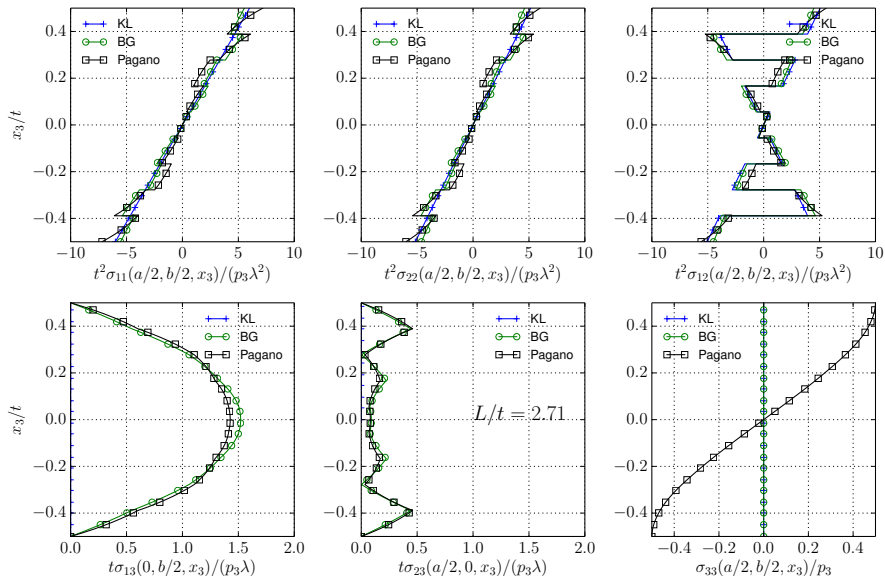
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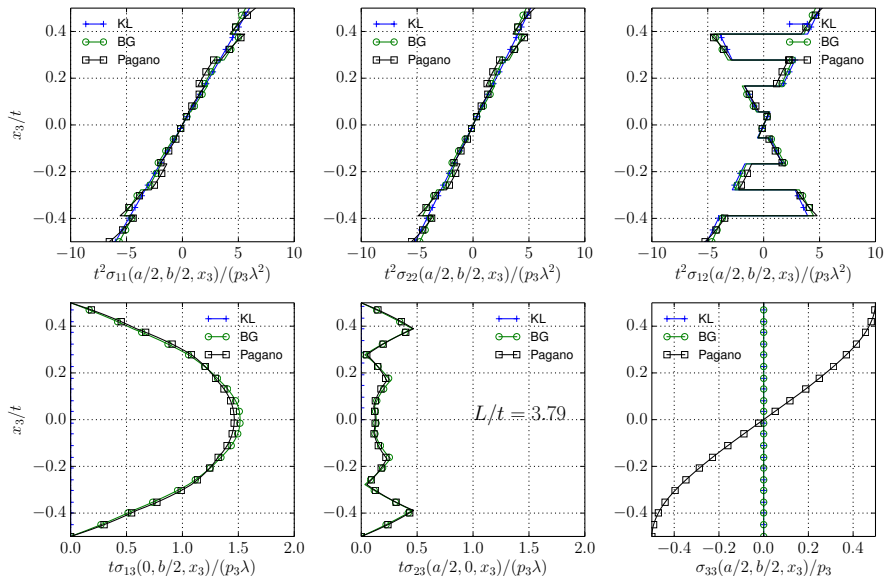
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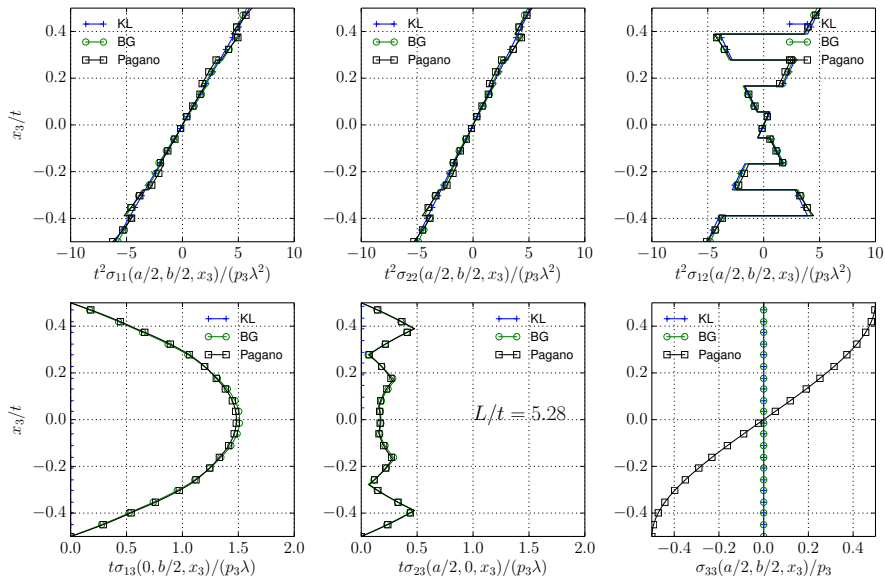
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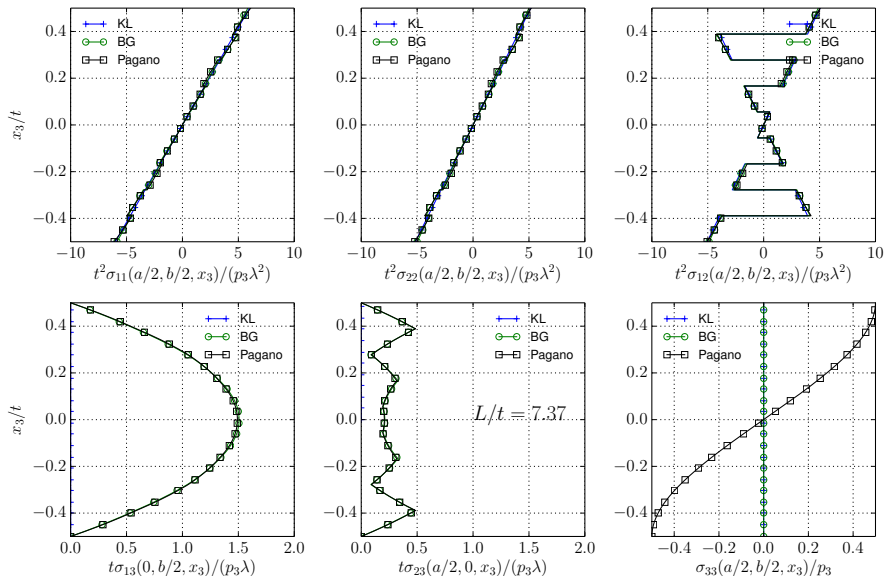
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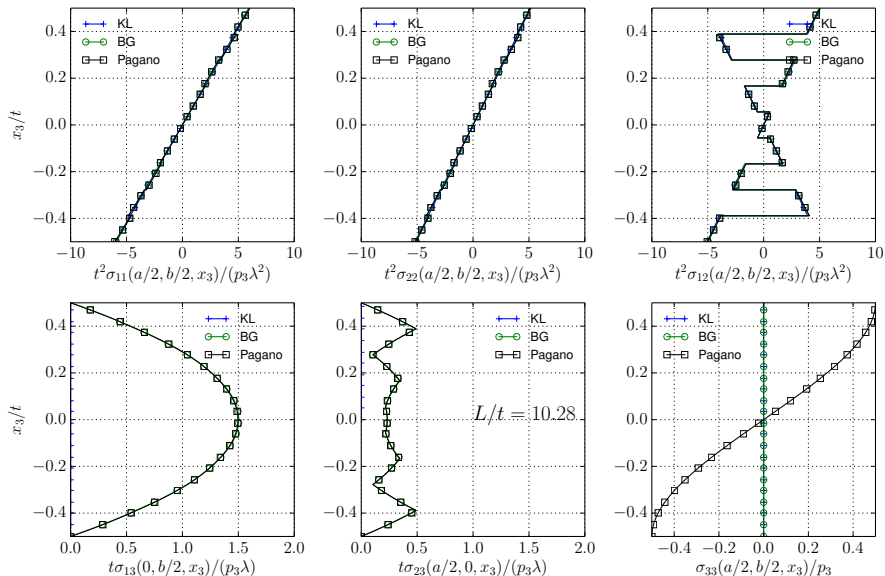
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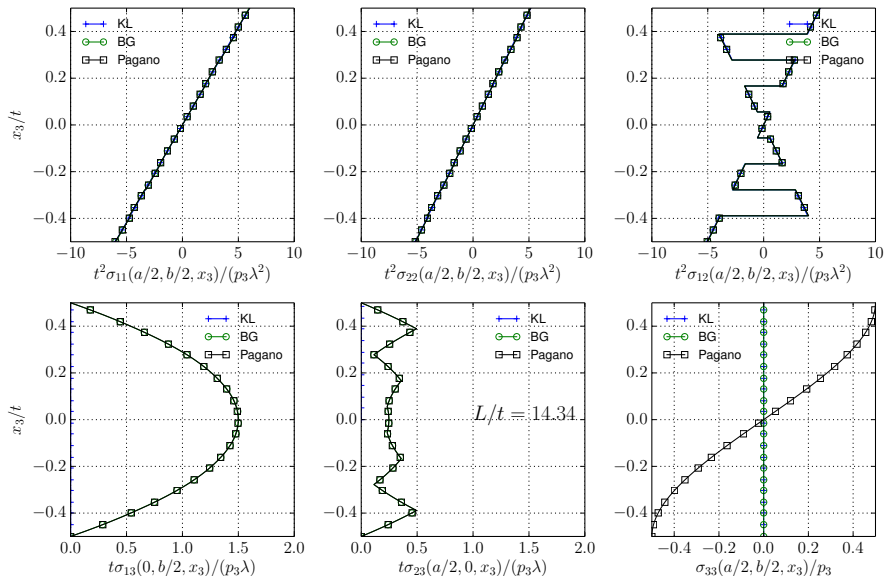
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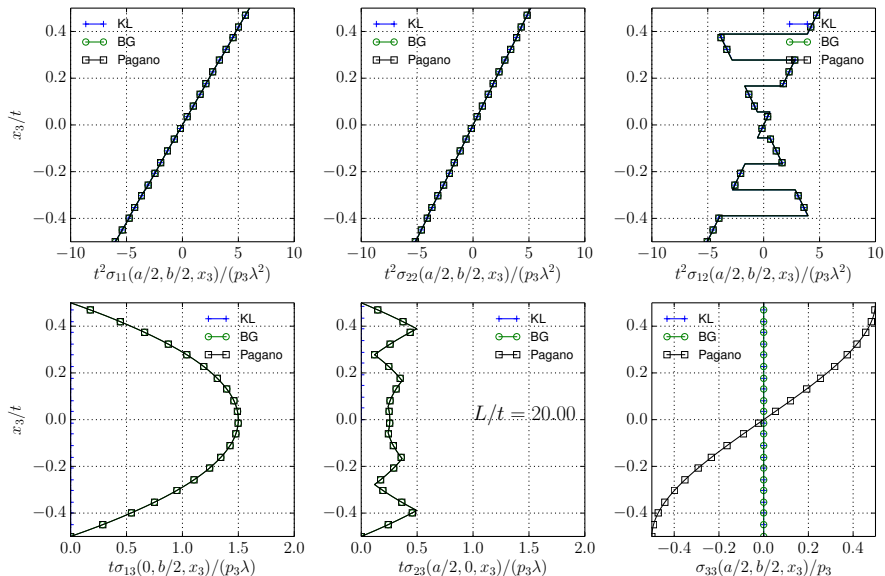
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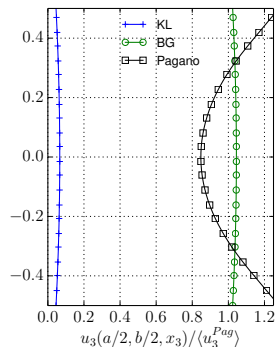
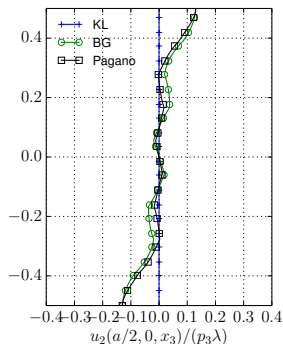
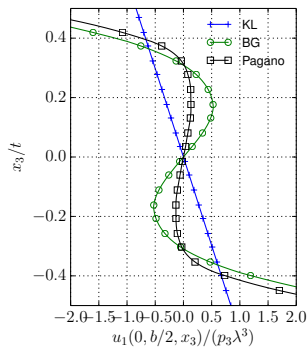
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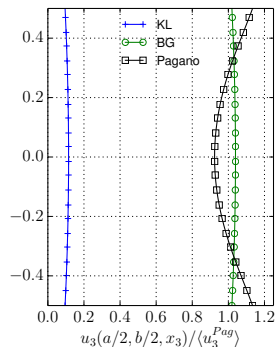
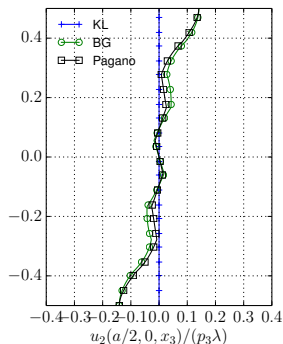
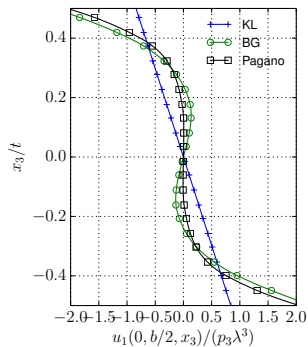
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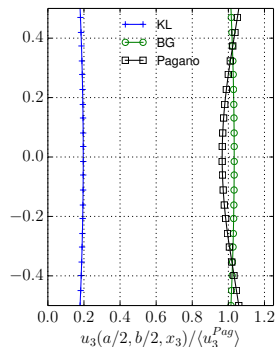
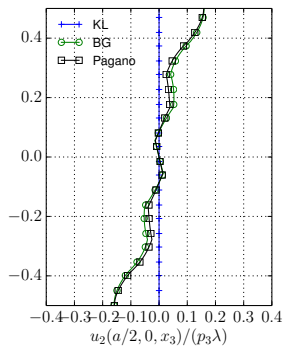
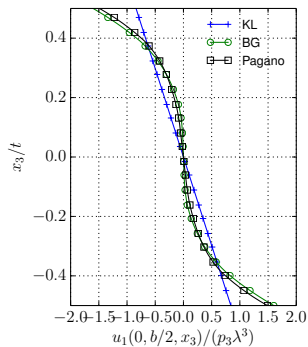
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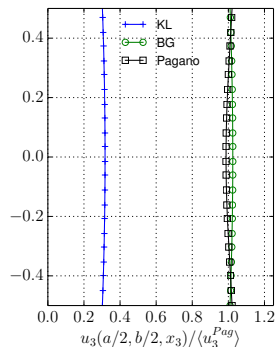
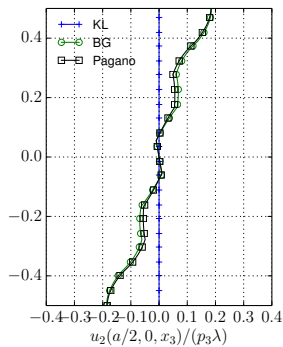
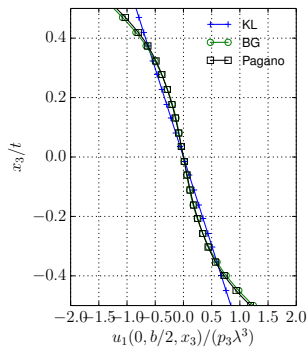
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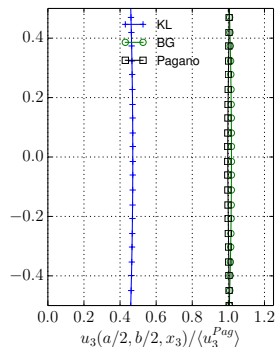
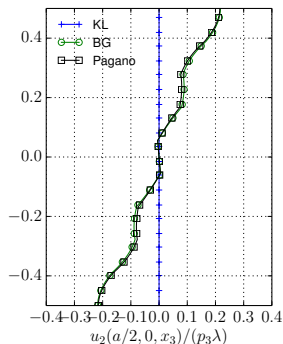
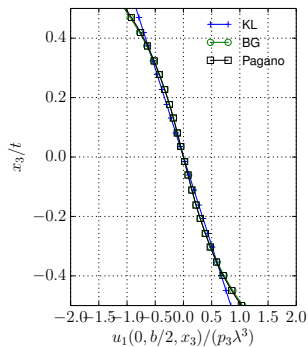
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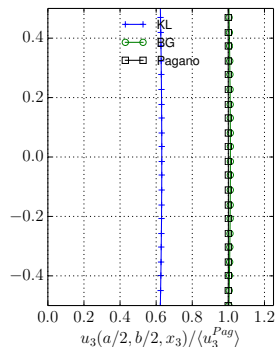
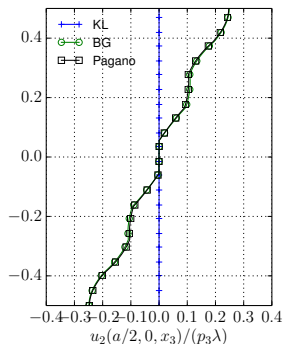
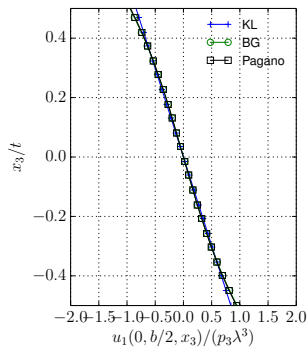
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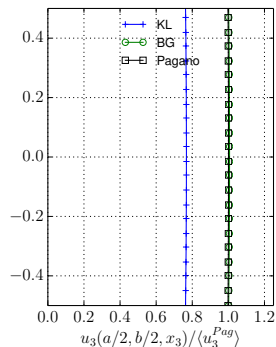
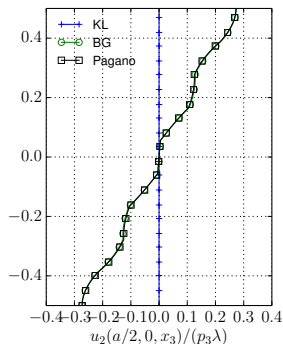
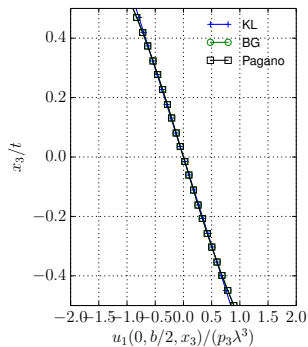
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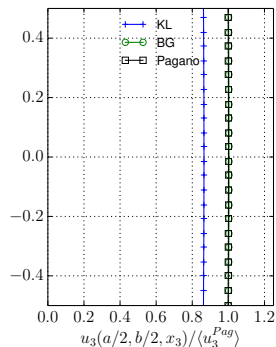
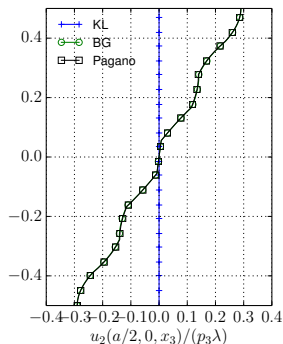
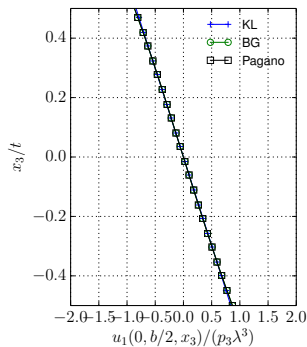
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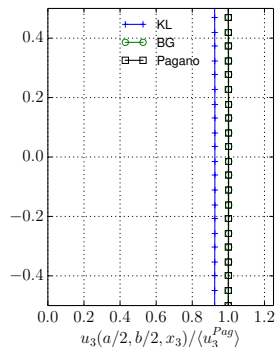
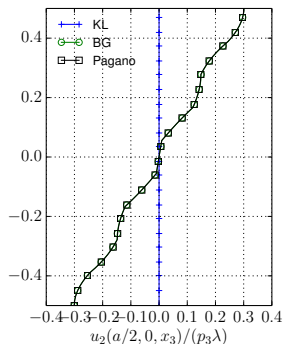
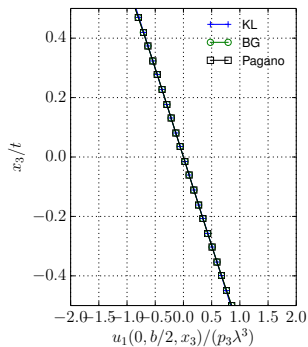
Displacement distributions for a $[45^\circ, -45^\circ]^4, 45^\circ$ stack



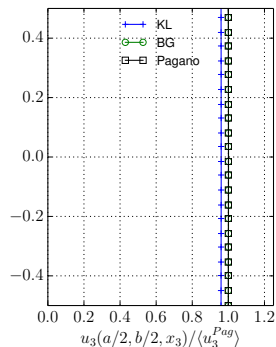
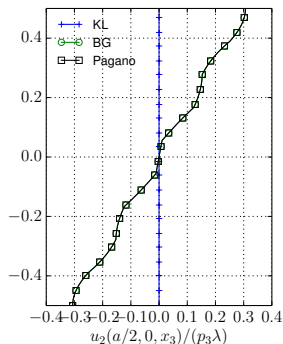
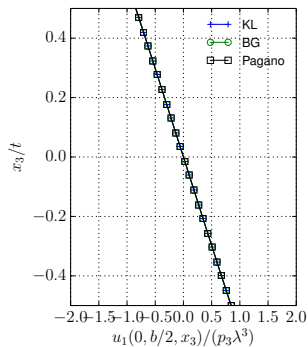
Displacement distributions for a $[45^\circ, -45^\circ]^4, 45^\circ$ stack



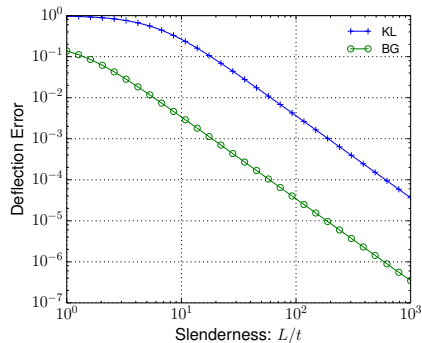
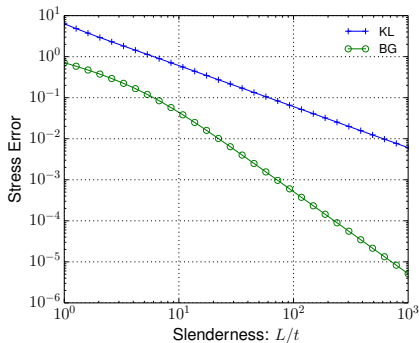
Displacement distributions for a $[45^\circ, -45^\circ]^4, 45^\circ$ stack



Displacement distributions for a $[45^\circ, -45^\circ]^4, 45^\circ$ stack



Convergence for a $[30^\circ, -30^\circ, 30^\circ]$ stack



$\Delta(\sigma)$ rate: $KL \sim t$ and $BG \sim t^2$

$\Delta(U_3)$ rate: $KL \sim t^2$ and $BG \sim t^2$

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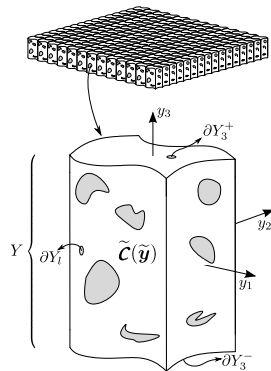
Extension to periodic plates

The case of cellular sandwich panels

Why all periodic plates are not “Reissner” like...

Extension to periodic plates

► Unit-cell and average estimates



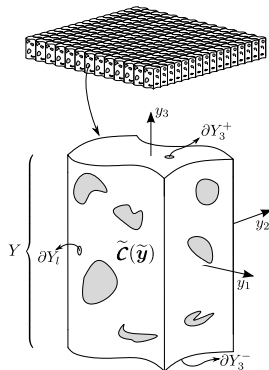
Extension to periodic plates

- ▶ Unit-cell and average estimates
- ▶ Bending auxiliary problem (Caillerie, 1984)

$$\mathcal{P}^K \left\{ \begin{array}{l} \tilde{\boldsymbol{\sigma}}^K \cdot \underline{\nabla} = 0 \\ \tilde{\boldsymbol{\sigma}}^K = \tilde{\mathbf{C}}(\underline{\mathbf{y}}) : \tilde{\boldsymbol{\varepsilon}}^K \\ \tilde{\boldsymbol{\varepsilon}}^K = y_3 \mathbf{K} + \underline{\nabla} \otimes^s \underline{\mathbf{u}}^{per} \\ \tilde{\boldsymbol{\sigma}}^K \cdot \underline{\mathbf{e}}_3 = 0 \text{ on free faces } \partial Y_3^\pm \\ \tilde{\boldsymbol{\sigma}}^K \cdot \underline{\mathbf{n}} \text{ skew-periodic on lateral edge } \partial Y_l \\ \underline{\mathbf{u}}^{per}(\underline{\mathbf{y}}) \text{ } (y_1, y_2)\text{-periodic on lateral edge } \partial Y_l \end{array} \right.$$

→ gives:

Localization $\underline{\mathbf{u}}^K$ $\tilde{\boldsymbol{\sigma}}^K$ related to the curvature \mathbf{K}
 Bending stiffness: $\tilde{\mathbf{D}}$



Extension to periodic plates

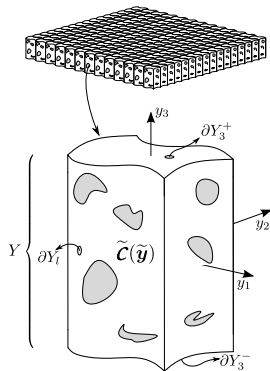
- ▶ Unit-cell and average estimates
- ▶ Bending auxiliary problem (Caillerie, 1984)
- ▶ Shear auxiliary problem

$$\mathcal{P}^R \left\{ \begin{array}{l} \underline{\sigma}^R \cdot \underline{\nabla} + \underline{\sigma}^M(\underline{y}) = 0 \\ \underline{\sigma}^R = \underline{\mathbb{C}}(\underline{y}) : (\underline{\delta} \otimes \underline{u}^M + \underline{\nabla} \otimes \underline{u}^R) \\ \underline{\sigma}^R \cdot \underline{e}_3 = 0 \text{ on free faces } \partial Y_3^\pm \\ \underline{\sigma}^R \cdot \underline{n} \text{ skew-periodic on lateral edge } \partial Y_l \\ \underline{u}^R(\underline{y}) \text{ } (y_1, y_2)\text{-periodic on lateral edge } \partial Y_l \end{array} \right.$$

→ gives:

Localization \underline{u}^R and $\underline{\sigma}^R$ related to \underline{R}

Shear compliance tensor: $\underline{\underline{f}}$



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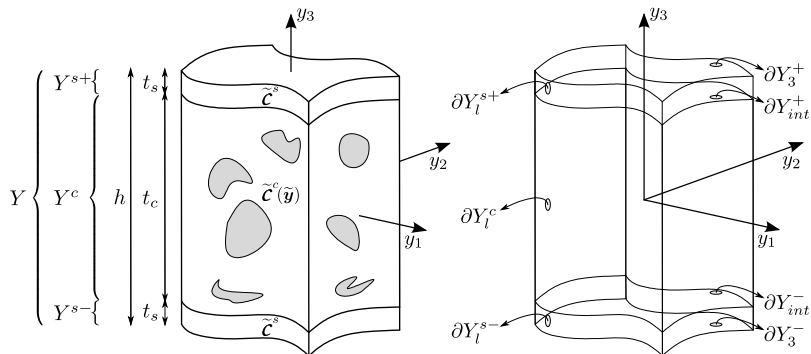
Extension to periodic plates

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Why all periodic plates are not “Reissner” like...

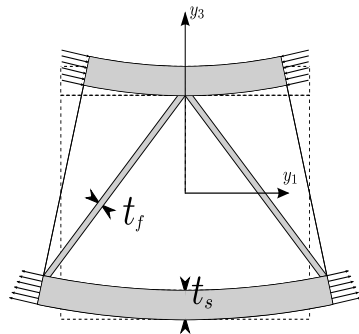
Justification of the Sandwich Theory

- Divide in 3 layers
(homogeneous skins and heterogeneous core)



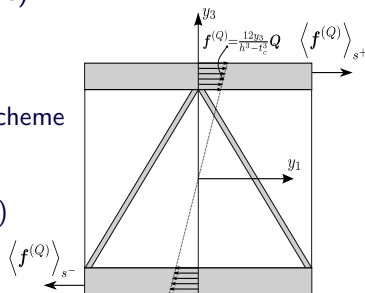
Justification of the Sandwich Theory

- ▶ Divide in 3 layers
(homogeneous skins and heterogeneous core)
- ▶ Bending auxiliary problem
 - ▶ Contrast assumption $\Leftrightarrow t_f \ll t_s$:
 $\rightarrow t_s/t_f$ Contrast ratio
- \Rightarrow Skins under traction/compression
- \Rightarrow Core not involved in Bending stiffness



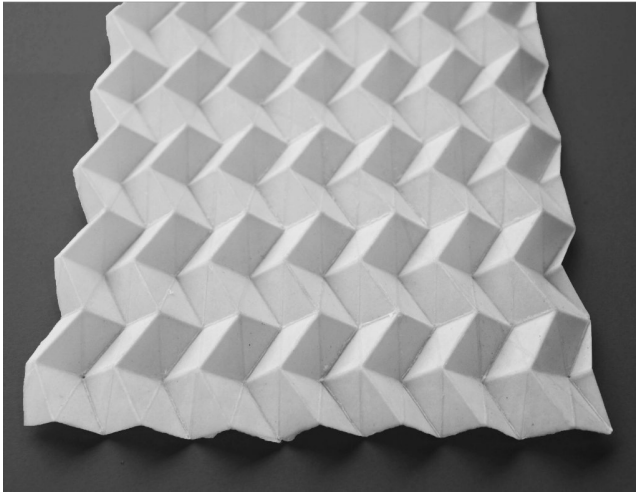
Justification of the Sandwich Theory

- ▶ Divide in 3 layers
(homogeneous skins and heterogeneous core)
- ▶ Bending auxiliary problem
- ▶ Shear auxiliary problem
 - ▶ \underline{f}^R becomes $\underline{f}^{(Q)}$ + Direct homogenization scheme
 - ▶ The BG is degenerated into RM model
 - ▶ $\underline{f}^{(Q)}$ confirms the classical intuition
 - ▶ Proof of the bounds from Kelsey et al. (1958)



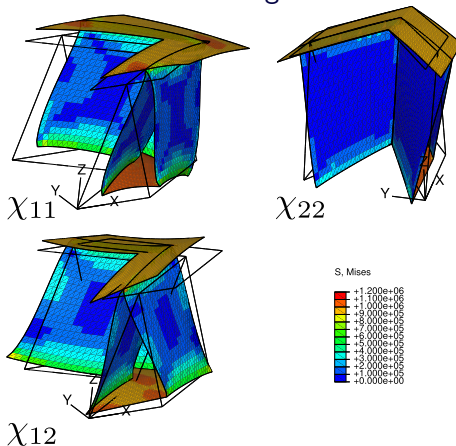
Lebée and Sab (2012a)

Application to the chevron pattern



Application to the chevron pattern

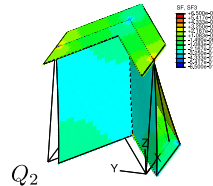
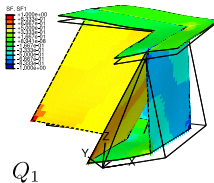
Bending:



Application to the chevron pattern

Shear forces
localization $\sigma^{(Q)}$

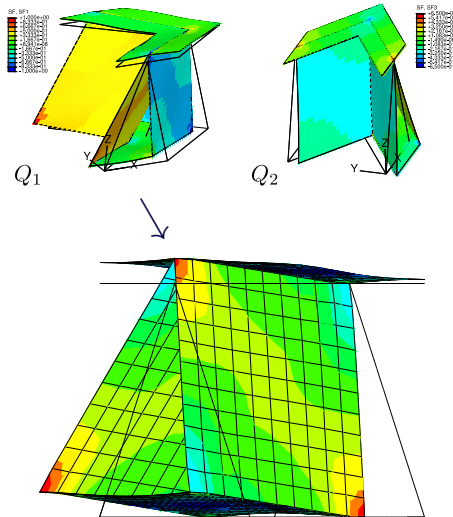
- Overall shearing
of the core



Application to the chevron pattern

Shear forces
localization $\sigma^{(Q)}$

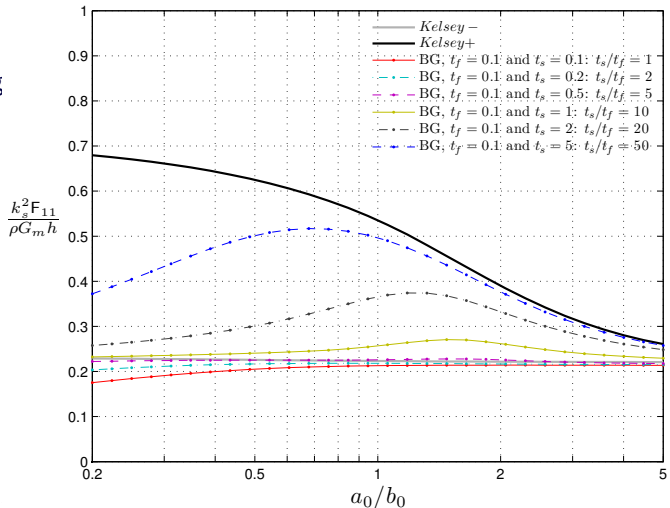
- Overall shearing of the core
- Out-of-plane skins distortion



Application to the chevron pattern

Shear forces
localization $\sigma^{(Q)}$

- Overall shearing of the core
- Out-of-plane skins distortion
- Critically influence shear force stiffness



Lebée and Sab (2012b)



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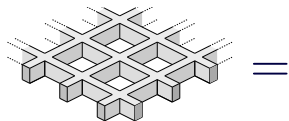
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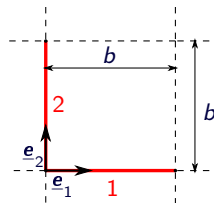
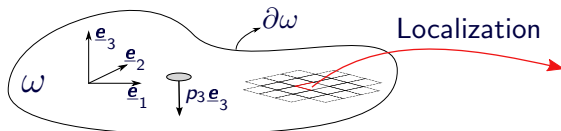
Homogenizing an orthogonal beam lattice?



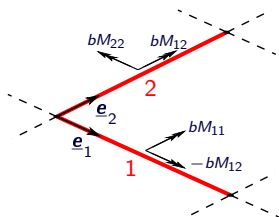
Thick-plate model (macro)



2 St-Venant Beams (micro)



Field localization



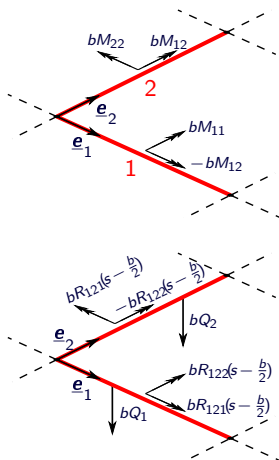
Bending moment ($\underline{r}^{(M)}, \underline{m}^{(M)}$):

Apply \underline{M} "on average" on the unit-cell (Caillerie, 1984)

$${}^1\underline{r}^{(M)} = {}^2\underline{r}^{(M)} = \underline{0}$$

$${}^1\underline{m}^{(M)} = \begin{pmatrix} -bM_{12} \\ bM_{11} \\ 0 \end{pmatrix}_1 \quad \text{and} \quad {}^2\underline{m}^{(M)} = \begin{pmatrix} bM_{12} \\ bM_{22} \\ 0 \end{pmatrix}_2$$

Field localization



Bending moment ($\underline{r}^{(M)}, \underline{m}^{(M)}$):

Apply \underline{M} "on average" on the unit-cell (Caillerie, 1984)

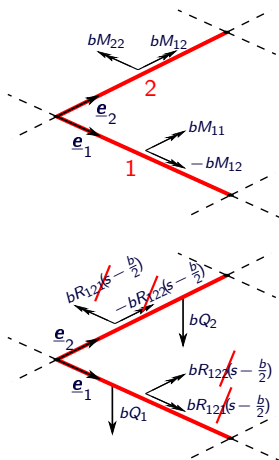
Bending gradient ($\underline{r}^{(R)}, \underline{m}^{(R)}$):

Assume $M_{\alpha\beta} = R_{\alpha\beta\gamma} X_\gamma$ (Lebée and Sab, 2013a)

$${}^1\underline{r}^{(R)} = \begin{pmatrix} 0 \\ 0 \\ b \underbrace{(R_{111} + R_{122})}_{Q_1} \end{pmatrix}_1 \quad {}^1\underline{m}^{(R)} = \begin{pmatrix} bR_{121} \left(s - \frac{b}{2}\right) \\ bR_{122} \left(s - \frac{b}{2}\right) \\ 0 \end{pmatrix}_1$$

$${}^2\underline{r}^{(R)} = \begin{pmatrix} 0 \\ 0 \\ b \underbrace{(R_{121} + R_{222})}_{Q_2} \end{pmatrix}_2 \quad {}^2\underline{m}^{(R)} = \begin{pmatrix} -bR_{122} \left(s - \frac{b}{2}\right) \\ bR_{121} \left(s - \frac{b}{2}\right) \\ 0 \end{pmatrix}_2$$

Field localization



Bending moment ($\underline{r}^{(M)}, \underline{m}^{(M)}$):

Apply \underline{M} "on average" on the unit-cell (Caillerie, 1984)

Bending gradient ($\underline{r}^{(R)}, \underline{m}^{(R)}$):

Assume $M_{\alpha\beta} = R_{\alpha\beta\gamma} X_\gamma$ (Lebée and Sab, 2013a)

Reissner-Mindlin ($\underline{r}^{(Q)}, \underline{m}^{(Q)}$):

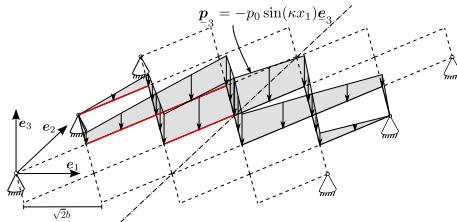
Assume cylindrical bending (Whitney, 1969; Cecchi and Sab, 2007)

$$Q_1 = R_{111}, \quad Q_2 = R_{222}, \quad R_{121} = R_{122} = R_{221} = R_{112} = 0$$

$${}^1\underline{r}^{(Q)} = \begin{pmatrix} 0 \\ 0 \\ bQ_1 \end{pmatrix}_1 \quad \text{and} \quad {}^1\underline{m}^{(Q)} = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix}_1$$

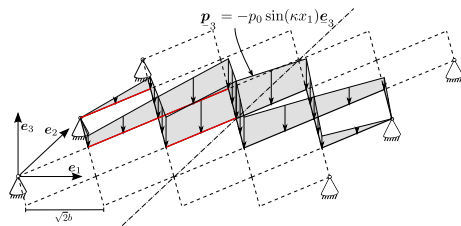
$${}^2\underline{r}^{(Q)} = \begin{pmatrix} 0 \\ 0 \\ bQ_2 \end{pmatrix}_2 \quad \text{and} \quad {}^2\underline{m}^{(Q)} = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix}_2$$

Application: lattice rotated 45° and cylindrical bending

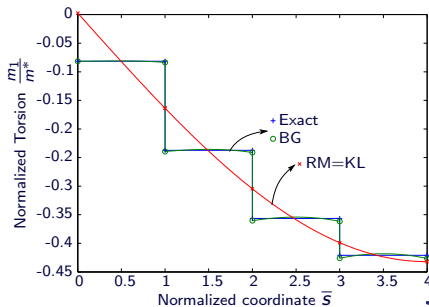
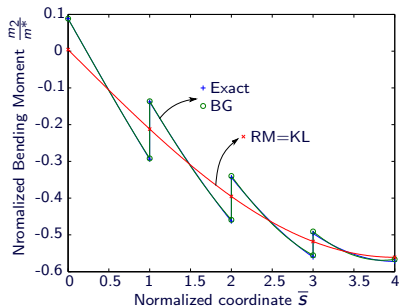


- Exact solution
- Plate solution + Localization (RM and BG)

Application: lattice rotated 45° and cylindrical bending



- Exact solution
- Plate solution + Localization (RM and BG)



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